Bundesrepublik Deutschland Federal Republic of Germany

Der Beauftragte Authorized Representative





UNITED KINGDOM CIVIL AVIATION AUTHORITY

CONDITIONS OF PERMIT TO FLY NO PR 051885/001

Daled: 19 JANUARY 2001

	Nationality and Registration Marks	Constructor and Constructor's Designation of Aircraft	Aircraft Serial Number	
	G-BZFV	DONNELLY IM ZENAIR CH 601UL	PFA 162A-13547	
1.	The eircraft shall not be exhibition or public demo	nown for the purpose of public transport or avrial work except aerial work which consists of onstration, including practice flights, test flights, and positioning flights associated with such de	flights for the purpose of public emonstrations.	
2.	The aircraft shall not be	fown over any congested and a city, town or settlement.		
3.		t designed for disassembly and re-assembly by the pilot prior to flight, the pilot in command s een made in accordance with the manufacturer's/PFA instructions.	that be satisfied that correct re-	
4.	The aircraft shall be main	ntained in an almosthy condition and in compliance with relevant Mandatory Permit Directives	h.	
5.	The aircraft shall be operated in accordance with the procedures and limitations contained in the manufacturers' instructions or in compliance with the associated Conditions of the Permit to Fly and PFA Operating Limitations. Where relevant the Conditions of the Permit to Fly and PFA Operating Limitations override the manufacturers' instruction.			
б.	No attenations or modifications shall be made to this aircraft or to its engines, propellers, or equipment, unless approved by the CAA or other Organisations approved by the CAA for the purpose.			
7.	A permanent placard shall be affixed to the aircraft in full view of the occupants, and shall be worded as follows:			
	() This aircra	Occupant Warning It has not been certificated to an International Requirement		
	(ii) Classificat	ion: Small Light Aeropiane		
8.	persons authorised by th	ate shall be issued certifying that the aircraft has been inspected and is fit for flight. This Cr te CAA. The Certificate shall be re-issued after overhaut, replacement, maintenance, repairs a is not required after pilot maintenance for those items listed in document reference PFAPM	modifications or inspections. A	

No entries or endorsaments may be made on this Certificate except by an authorised person.

CA960PFA/SLA/C 210200 POPULAR FLYING ASSOCIATION

Airworthiness Approval Note: PFA-162A-420

Aircraft Type: Zenair CH601 UL

Serial Number: PFA 162A-13585 and PFA 162A-13547

Registration: G-ZODI and G-BZFV

Foreword

The Zenair CH601 UL is an 'SLA' variant of the Zenair CH601 HD, the latter having been accepted by the PFA for many years as a 'group A' homebuilt aircraft, several examples of this type having been built under PFA supervision.

G-ZODI and G-BFZV are the firs UK examples of the CH601 UL variant to be completed and flown, differing only in their choice of engine and propeller.

This AAN has been raised as a design reference for G-ZODT and G-BZFV and any subsequent examples of the CH601 UL built to the same standard.

Issue 1

Initial issue

19th January 2001

Approved:

Dated: 19th January 2001

For the Popular Flying Association CAA Approval DAI/1172/48 AAN PFA-162A-420 Page 2 of 6

PFA TYPE DATA SHEET

Aircraft Type: Zenair CH601 UL

Engine Type: Rotax 912-UL (G-ZODI) or Rotax 912-S (G-BZFV)

Propeller Type: GSC Tech-III 68" diameter x 13 degrees (G-ZODI) or Sport Prop VP 2 bladed 1660MM diameter (G-BZFV)

LIMITATIONS

1. Maximum number of occupants authorised to be carried: Two

 The aircraft must be operated in compliance with the following operating limitations, which shall be displayed in the cockpit by means of placards or instrument markings:

2.1 Aerobatic Limitations This aeroplane is permitted to fly only for nonaerobatic operation. In this context non-aerobatic operation includes: i) Any manoeuvre necessary for normal flying ii) intentional stalls from level flight iii) steep turns in which the angle of bank does not exceed 60 degrees Intentional spinning is prohibited.

- 2.2 Loading limitations Maximum Total Weight Authorised: 450 Kg CG Range: 10.5 inches to 17.5 inches aft of datum. Datum Point is: The leading edge of the wing.
- 2.3 Engine Limitations

Maximum Engine RPM: 5800 Maximum Continuous Engine RPM: 5500

2.4 Airspeed Limitations

DESCRIPTION

The Zenair CH601 UL is a conventionally configured low wing cantilever monoplane aircraft of rivetted aluminium alloy construction, seating two side by side in an enclosed cockpit. It is available as a kit or as a set of plans from European Zenair sub-contract manufacturers the Czech Aircraft Works. Both G-ZODI and G-BZFV have been built from quick-build kits supplied by Czech Aero Works. The kit may be purchased in two formats which are considered to meet the 500 hour amateurbuilding criteria of CAP 659, being the guick-build kit and the 49% kit options. The 100% kit available from Czech Acro Works is not eligible in the UK.

The CH601 UL model is a variant of the CH601 series which has been developed for the European ultralight market. Differences between the CH601 UL model and the already-accepted CH601 HD are as follows:

- a. Addition of three-piece split flaps to lower the stall speed. These are of rivetted aluminium alloy construction, and operated by a mechanical system of torque tubes, pushrods and levers, and actuated by a hand lever on the port cockpit sidewall.
- ь. The structural reinforcements which distinguish the CH601 HD 'heavy Duty' model from the original CH601 model have not been included.

As with the CH601 HD, the aircraft is available with several alternative options including the choice of nosewheel or tailwheel undercarriage and a side-opening or front-opening canopy. G-ZODI is fitted with the nosewheel undercarriage and side-opening canopy, while G-BZFV is fitted with a nosewheel undercarriage and forward-opening canopy.

A number of different fuel tank configurations are available, including a front fuselage tank, wing tanks between the wing spars and wing tanks in the wing leading edges. Both G-ZODI and G-BZFV are fitted with a fuselage tank only, in order to keep the aircraft's empty weight within permitted limits, and it is unlikely that additional fuel tanks will be able to be accepted on the CH601-UL model in the UK.

A number of different engine options are available including the 80 BHP Rotax 912-UL and 100 BHP Rotax 912-S. G-ZODI is fitted with the Rotax 912-UL engine while G-B2FV has the uprated 912-S.

When fitted with the Rotax 912-UL and 912-S engine, the Zenair option of moving the firewall forward by 50mm is considered essential to keep the aircraft's empty cg forward and allow safe loading within the reduced (17.5" as opposed to the manufacturer's recommended 19.7") aft og limit specified by the PFA.

PFA-162A-420 Page 4 of 6

The standard aileron hinging rely on flexure of the top wing skin for aileron deflection. This is considered acceptable although some owners have found that this causes what they considered excessive _ileron control forces, and have fitted the optional conventional piano type aileron hinges. This option is also considered acceptable by the PFA.

BASIS FOR APPROVAL

The Zenair CH601 UL has been assessed as a variant of the already accepted Zenair CH601 range using BCAR Section S issue 2 as the reference design code where appropriate.

INVESTIGATION

(1) It has been checked that empty weights of G-ZODI and G-B2FV at 579 Lbs and 578 Lbs respectively are less than the maximum permissable empty weight as specified by BCAR S23, S25 and S29 when operated as a two seater with a maximum gross weight of 450 Kg. Assuming a maximum cruise fuel consumption of 10 Kg/hr the maximum permissible empty weight is 450-86-86-10 = 268 Kg (589 Lbs)

(2) A stress analysis of the CH601 airframe has been submitted showing compliance with all the main requirements of FAR 23 subpart C and D at the higher maximum gross weight of 1058 Lbs (489 Kg). This is considered to provide at least equivalent safety with complying with BCAR Section S subpart C and D at a maximum gross weight of 450 Kg (992 Lbs).

(3) A separate stress analysis of the split flap installation has been provided showing compliance with the applicable paragraphs of BCAR Section S Subpart c and D. The analysis of the flap operating system resulted in the flap operating lever having to be replaced a reinforced operating lever of 4" x 0.058" wall 4120N in order to meet pilot effort requirement of \$405.

(5) Zenair have submitted a set of pilot notes for the aircraft including a supplement dealing with the use of the split flaps.

MODIFICATION STATE

The following UK modifications have been called for in order to provide an adequate level of safety:

MOD/162A/001 Flap operating lever of increased wall thickness (3/4" x 0.058" 4130N fitted, to comply with BCAR S para S405. Wall thickness on original lever 0.035". Czech Aircraft Works supplied the new component.

MOD/162A/002 Firewall moved forwards 50mm to improve aircraft's empty og position. Czech Aircraft Works optional 'Extended Fuselage Kit' refers.

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MOD/162A/003 Elevator and rudder control systems must exhibit minimum control friction in order to ensure an acceptable level of pitch and yaw stability.

MOD/162A/004 Nosewheel steering to be provided with additional return springs to provide adequate level of directional stability. Drawing MOD/162A/004 refers.

MOD/162A/005 In common with the CH601 HD series aircraft, Conventional aircraft turnbuckles of rating appropriate to cable diameter substituted for 'home made' cable tensioners in flying control system. Barrel p/n AN155-126S/1999 and eye ends MS2125/C3RS and AN170/16LS refer.

MOD/162A/006 In common with the CH601 HD series aircraft, Either the aircraft must be fitted with a forward-opening canopy option (drawings CF-1 to CF-14 refer) or if the standard sideward opening canopy is fitted, the inner canopy opening levers must be omitted and the shoulder harnesses prevented from being able to get caught in the rear canopy retaining claw, a suitable method being by fitting a short length of split foam domestic water pipe insulation sleeve over the harness in the area concerned, retained by tie-wraps.

MOD/162A/007 Aft cg limit restricted to 17.5" AOD. In order to achieve this, any significant weight of baggage must be carried in the wing baggage lockers rather than the rear cockpit locker.

PLACARDS

In addition to the placards called for on page 2 this AAN, the following placard must also be fitted in the cockpit in full view of the occupants: (missing figures to be completed as appropriate)

Category: Small Light Aeroplane

Empty weight: ? Kg date of weighing ? Max Total weight Authorised: 450Kg Maximum permitted empty weight: 268 Kg Maximum fuel load with two occupants of 86 Kg each, ? litres Maximum combined crew weight with full fuel, ? Kg

FLIGHT TEST

G-ZODI and G-BZFV have both completed in excess of twelve hours flying without defect and been flight tested to PFA flight test schedule 81/1/1.

G-B2FV has been the subject of an independent flight test under PFA supervision. Performance and handling has been found to be essentially 'identical to previous examples of the CH601 variants fitted with Rotax 912 engines, being characterised by

PFA-162A-420 Page 6 of 6

powerful lateral stability but weak but just positive longitudinal static stability and directional static stability, within the restricted aft cg limit of 17.5" AOD. Control forces are light about all axes and the aircraft is easy to fly despite the light stabilities in yaw and pitch. Stall warning is particularly noticeable due to drumming of the rear fuselage skins in the buffet, and the aircraft displays a particularly benign stall.

Operation of the wing flaps deployed has been found to be satisfactory, the flaps being easily operated and giving a useful steepening of the glide angle, with associated greater nose-down pitch attitude. The stall speed with flaps fully deflected was measured using a GPS, averaging the results of a series of stall tests carried out upwind and downwind, and found to be 40 mph, placing the aircraft into the SLA category. The stall characteristics with flaps deployed are similarly benign, producing slightly greater wing dropping tendency but still gentle and controllable with ailerons and rudder still functional throughout. Flight test results are held in file PFA/G-BZFV.

NOISE CERTIFICATION

Noise certification flight tests have been carried out successfully on G-ZODI and G-BZFV, Noise Certificates #1656 and 1652 issued accordingly on 22.9.2000 and 14.9.2000 respectively.

RECOMMENDATION

It is recommended that the aircraft is issued with a Permit to Fly subject to the limitations shown on page 2. It is certified that the aircraft complies with the stated design standard and is built to accepted aeronautical standards.

Date: 19th January 2001 Approved:

For the Popular Flying Association CAA Approval Number DAI/1172/48

SAFETY REGULATION GROUP

Avlation House	Direct Dial	01293 573179	Switchboard	01293 567171
Gatwick Airport South	Direct Fax	01293 573972	Fax	01293 573999
West Sussex			Telex	878753
RH6 OYR				

2 5 JAN 2001

Mr J M Donnelly Birse Lodge Cottage Charleston Road Aboyne AB34 5EL

Our ref 9/23/G-BZFV

24 January 2001

Dear Sir

REGN:- G-BZFV TYPE:- ZENAIR CH 601UL SERIAL NO.:- PFA 162A-13547

Following a recommendation from the Popular Flying Association for the above mentioned aircraft, please find enclosed

Documents enclosed:

Permit to Fly PFA Operating Limitations Certificate of Validity

Yours faithfully

JL.

Joan Lacey (Mrs) Applications & Certification Section



TYPINTYG

Nr UL B-35

Detta typintyg intygar att nedan angiven typ av ultralätt flygplan uppfyller luftvärdighetsfordringarna i Bestämmelser för Civil Luftfart (BCL), BCL-M 5.4

Produkt	Ultralātt flygplan klass B av typ ZENAIR CH 701 STOL			
Ansökan	ELVO ELECTRONIC ARCHAID SWEDEN AB, Viktoriagatan 15, 411 25 GÖTEBORG genom Motorflygförbundet KSAK (L2000- 2648-1041)			
Typunderlag	Sammanställd teknisk specifikation UL B-35, upprättad av Motorflygförbundet KSAK den 16 augusti, 2000 (L2000-2648- 1041)			
Begränsningar	Enligt specifikation UL B-35 samt däri specificerad flyghandbok eller senare revision			
Referens för godkännande	Ansökan om tillstyrkan om godkännande i brev från Motorflygförbundet KSAK daterat den 16 augusti, 2000 (L2000- 2648-1041)			
Norrköping den 23 augusti, 2000				
LUFTFARTSVERKET Luftfartsinspektionen				
Sul	En ility Weak			

Eskil Wiklund

NAV May

Nils von Koch

Translation from the origin TYPE ACCEPTANCE CERTIFICATE N0. UL B-35

This Type Acceptance Certificate confirms that the type of ultra-light aeroplane specified below complies with the airworthiness requirements in Bestämmelser för Civil Luftfart (BCL), BCL-M 5.4

Product	Ultralight aeroplane class B of type ZENAIR CH 701 STOL
Application	ELVO ELECTRONIC ARCHAID SWEDEN AB, Viktoriagatan15, 411 25 GÖTEBORG through Motorflygförbundet KSAK (The Royal Swedish Aeronautical Aeroclub) (L2000-2648-1041)
Type Design	Technical Specification UL B-35, drawn up by Motorflygförbundet KSAK August 16, 2000 (L2000-2648-1041)
Limitations	According to Technical Specification UL B-35 and the Aeroplane Flight Manual specified in that Technical Specification or later revision
Approval References	Application for recommendation of approval in a letter from Motorflygförbundet KSAK dated 16, 2000 (L2000-2648-1041)

Norrköping August 23, 2000 LUFTFARTSVERKET Luftfartsinspektionen

Nitvhas

Eskil Wiklund

Nils von Koch

SLOVENSKÁ REPUBLIKA LETECKÝ ÚRAD

SLOVAK REPUBLIC CIVIL AVIATION AUTHORITY

OSVEDČENIE LETOVEJ SPÔSOBILOSTI

CERTIFICATE OF AIRWORTHINESS

0564 - S

č./No:.....

Značky štátnej príslušnosti a registra : Nationality and Registration Marks :	Výróbca a typ lietadla : Manufacturer and Designation of Aircraft :	Výrobné číslo lietadla : Aircraft Serial Number :
OM-ATIS	ATIS s.r.o., letisko Slávnica ZODIAC CH-601	6 - 2212
	Jltraľahký letún - Súkromná Jltralight aeroplane - Private	

Toto osvedčenie letovej spôsobilosti je vydané v zhode s Dohovorom o medzinárodnom civilnom letectve zo dňa 7.12.1944 a podľa zákona zo dňa 24.9.1956 č. 47 Zb. o civilnom letectve a predpisov na jeho základe vydaných pre uvedené lietadlo, ktoré sa považuje za spôsobilé na lietanie, ak bude udržované a v prevádzke používané podľa príslušných predpisov s príslušnými prevádzkovými obmedzeniami.

This Certificate of Airworthiness is issued pursuant to the Convention on International Civil Aviation dated 7° December 1944 and to the Civil Aviation Act No. 47 dated 24° September 1956 and regulations made thereunder in respect of the above-mentioned aircraft which is considered to be airworthy when maintained and operated in accordance with the pertinent regulations and operating limitations.

2 3. SEP. 1998

Dátum vydania / Date of issue



Podpis / Signature

SLOVENSKÁ REPUBLIKA LETECKÝ ÚRAD

SLOVAK REPUBLIC CIVIL AVIATION AUTHORITY

ZVLÁŠTNE OSVEDČENIE LETOVEJ SPÔSOBILOSTI SPECIAL CERTIFICATE OF AIRWORTHINESS

17/2000

č.: _____ No.

Značky štátnej príslušnosti a registrové značky: Nationality and Registration Marks:	Výrobca a ty Manufacture Aircraft:	yp lietadla: r and Designation o	Výrobné číslo: f Aircraft Serial №:
ATIS		S, spol.s r.o LÁVNICA – 601 HDS	6 - 3687
Vydané na základe: Basic of issuance: Letecké	ho predpisu L	8/A ods. 4.3.4.1. pi	sm. d)
Vydané pre: Issued to: ATIS, spol.s r.o. letisko 018 54 SLÁVNICA			obné letové skúšky ovyrobeného lietadla
Obmedzenia: Limitations: - Dané programom letových skúšok a letovou príručkou - Pilot: Karol Benedikovič, č.p. 02980030 Milan Rehák, č.p. 02970148		Platné pre oblasť/trať: Valid for area/line: územie Slovenskej republiky	
Dátum vydania: Date of issuance: 17.04. 2000			oo do vykonania sériovej ho čo nastane skôr
Riaditeľ LÚ SR Director CAA SR	Pečiatka: Stamp: 51.0	VENSK	odpis: gnature:
Ing. Ján KAŠŠÁK	STITCKY ,	JULL -	Laure

SLOVENSKÁ REPUBLIKA LETECKÝ ÚRAD

SLOVAK REPUBLIC CIVIL AVIATION AUTHORITY

OSVEDČENIE O ZÁPISE LIETADLA DO REGISTRA LIETADIEL

CERTIFICATE OF REGISTRATION

Ċ.:.... 0564-S No.

1. Značky štátnej príslušnosti	2. Výrobca a typ lietadla:	3. Výrobné číslo lietadla:	
a registra: Nationality and Registration Marks:	Manufacturer and Designation of Aircraft:	Aircraft Serial Number:	
0 M - A T I S	ATIS s.r.o. letisko Slávnica ZODIAC CH-601	6-2212	
4. Názov vlastníka Name of owner Názov prevádzkovateľa	ATIS s.r.o. SNA gen.M.R.Štefánika	-	
Name of operator 5. Adresa vlastníka			

6. Týmto sa osvedčuje, že hore uvedené lietadlo bolo riadne zapísané do REGISTRA LIETADIEL SLOVENSKEJ REPUBLIKY

v súlade s Dohodou o medzinárodnom civilnom letectve zo dňa 7. decembra 1944 a podľa ustanovenia zákona o civilnom letectve.

It is hereby certified that the above described aircraft has been duly entered on the AIRCRAFT REGISTER OF SLOVAK REPUBLIC

in accordance with the Convention of International Civil Avlation dated 7th December 1944 and with the provision of the Law on Civil Aviation.



Podpis / Signature

23.9.1998

Dátum vydania / Date of issue

Letecka umaterská asociace ČR - Linghi Alternaft Association of Czech Republic

Typový průkaz – Type Certificate

Letecká Amatérská Asociace české republiky (dale LAA CR) v souladu s ustanovením 8/81 odst. 2. zákona o evvilním letectví č. 49/1997/Sb. byla dne 1/4/1997 pověřena Uradem pro civilní letectví vymeženými činnostmi, které souvisejí s tvrobou a provozem sportovních letajících zárížení (dále SLZ) v České republice, c. j. : 1539/Pl-165/97

> Označení typu letecké techniky: Designation of the type of aviation technology:

Dvoumístný, jednomotorový, aerodynamicky řízený, celokovový dolnoplošný ultralehký letoun - sportovní létající zařízení Typové označení: ZENAIR CH 601 UL ZODIAC Maximální vzletová hmotnost 450 kg. Podrobné technické specifikace jsou uvedeny v příloze.

> Držitel typového průkazu: Owner of Type Certificate:

CZECH AIRCRAFT WORKS, s.r.o. Luční1824 686 02 Staré Město

Schváleno technickou komisí LAA ČR dne: Approval of the Technical commission of LAA CR:

9.3.2001

Typový průkaz je zaregistrován u LAA ČR pod značkou: Type certificate is registered in LAA CR under registration sign:

ULL 07 / 2001

el technického odboru LAA Chief Engineer of LAA CR:	ČR:	Prezident LAA ČR: President of LAA CR:
	Letecké amatéroká asociace ČR Ke Kablu 289	· · · · · · · · · · · · · · · · · · ·
Ing. Václav Chvála	102 00 Praha 10 Tel.: 71085 274 6.4	Ing, Lubos Zahradník (

Letecká amatérská asociace ČR - Light Aircraft Association of Czech Republic

Typový průkaz - Type Certificate

Letecká amatérská asociace české republiky (dále LAA ČR) v soulada s ustanovením § 81 odst.2) zákona o civilním letectví č. 49/1997 Sb. byla dne 1.4. 1997 povětena Úřadem pro civilní letectví vymezenými činnostmi, které souvisějí s výrobou a provozem sportovních létajících zařízení (dále SLZ) v České republice, č.j. : 1539/PI-165/97

Označení typu letecké techniky : Designation of the type of aviation technology :

Dvoumistný, jednomotorový, aerodynamicky řízený celokovový ultralehký letoun typové označení ZENAIR CH 701 Maximálni vzletová hmotnost 450 kg. Podrobné technické specifikace jsou uvedeny v příloze. Letouny jsou vyráběny v licenci firmy Zenair, Huronia Airport, MIDLAnd Ont., Canada

> Držitel typového průkazu : Owner of Type Certificate :

Czech Aircraft Works. s.r.o. Luční 1824 686 02 Staré Město

Schváleno technickou komisi LAA ČR dne: Approval of the Technical commission of LAA CR:

16.7. 1996 Dne 8.11.2000 převedeno z firmy KAPPA na Czech Aircraft Works s.r.o.

Typový průkaz je zaregistrován u LAA ČR pod značkou: Type certificate is registered in LAA CR uder registration sign:

ULL - 17/96

Ředitel technického odboru LAA ČR :	
Chief Engineer of LAA CR:	

LL.

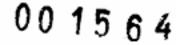
Ing. Vactav Chvála

Letecká amatérská asociace ČR Ke Kablu 289 102 00 Praha 10 Tel. 707 52 70 č.3

Prezident LAA ČR : President of LAA/CR

Ing Lub6





PORMA LA CERT

SECRETARIA DE COMUNICACIÓNES

Y

TRANSPORTES

DIRECCION GENERAL DE AERONAUTICA CIVIL DIRECCION TECNICA Y DE SUPERVISION SUBDIRECCION DE INGENIERIA DEPARTAMENTO DE INGENIERIA AERONAUTICA

CERTIFICADO DE APROBACION TIPO No. IA-0494

Le Secretería de- Comunicaciones y Transportes, en base al Artículo 17, Fraccion XIV de su Reglamento interno y por conducto de la Dirección General de Aeronáutica Civil, con fundamento en los Artículos 318 y 368 de la Ley de Vías Generales de Comunicación, en vigor, expide el presente documento a favor de:

PUBLICIDAD DE ALTURA, S.A. DE C.V.

certificando que lo enunciado a continuación reúne las especificaciones mínimas aplicables para su diseño, fabricación, utilización y operación segura, en acuerdo a las Normas, Procedimientos y Regulaciones prescritas por la Dirección General de Aeronáutica Civit:

AERONAVE LIGERA MOTORIZADA

MARCA : ZENITH AIRCRAFT COMPANY MODELO: STOL CH701 No. DE SERIE: 7-3091 MATRICULA: XB-FXX

VIGENCIA: Este Certificado tiena validez indefinida a no ser que sea Cancelado, Suspendido o Revocado por la Autoridad Aeronáutica correspondiente, y no es válido sin tas hojas de especificaciones anexas.

Fecha de emisión

22 - JULIO - 1994

Atentamente SUFRAGIO EPECTIVO, NO REA LEOCION EL DIRECTOR GENERAL IOT FEDERICO CANO AS THE

Qualquier eltereptión a sete Certificado será sancionado con todo el rigor que corresponda, de ecuardo e lo establecido en la Lay de Vies Generales de Comunicación y sus Regismentos. STATE OF ISRAEL MINISTRY OF TRANSPORT Civil Aviation Administration



מדינת ישראל משרר את הבורה מינהל התעופה האזרחיה

תעורת סוג לכלי טיס TYPE CERTIFICATE

No. 11-124 .00

MAKE: ZENAIR LID., P.O.BOX 235, MIDLAND, DN., I.AR 4KB CANADA

REPRESENTED BY; CONDOR AVIATION

This certificate issued to <u>34 ARLOZOROU ST., RAMAT-GAN, ISKAL</u> certifies the: the Type Design for the following product with the openating limitations and conditions therefor as specified in the Ayistion Regulations (Certification procedures for oncieft and parts thereal) 1677 and the Type Certificate Data Sheet, meets Type Certificate Data Sheet, meets the alworthiness requirements of TP 10141, OESIGN STANDARD FOR ADVANCED ULTRALIGNT AEROPLANES (CANADA)

MODEL: ZENAIR STOL CH 701 (ULTRALIGHT)

This certificate and the Type Certificate Data Sheet which is a part thoreof shall remain in effect until sourcendered, suspended, revoked, or a termination data is otherwise established by the lifed of the Civil Aviation Aduphistration.

מעווח זו וגליון הנתובים של יועררת הסוג, המחווח חלק בלתי נכור שטטו. תהיון במוקף עד ביטולה, התליחת, חשעיונה או עד פקיעת תוקפון כפי שנקבע כה על ידי דאש סינותל המעופה האזרחית.

Date of application 5.05,1998 nerve nerve

Date of issuance 23.11.1993 חואריך הוצאת התעורה

בחתאם לחוויאות ראש מינהל התעופה האזרתיוו

By the direction of the Head of the Civil Avialion Administration

Signature J. GILEAD, T. G. התימת THE MANAGER, ENG. & MEG. BRANCH GAA) 20110

and the second state of th

Pursuant to Subsection 5.9(2) of the <u>Aeronautics Act</u>, I hereby authorize the operation of Zenair, models Zodiac CH 601 and STOL CH 701 aeroplanes in Canadian airspace without complying with the requirement to hold a Certificate of Airworthiness as required by paragraph 210(1)(a) of the <u>Air Regulations</u>, subject to the following conditions:

- the aeroplane shall meet and be maintained to the standards published in TP 10141, Design Standards For Advanced Ultra-Light Aeroplanes;
- (b) the aeroplane shall be operated in accordance with all <u>Air Regulations</u> and <u>Air Navigation Orders</u> relating to ultra-light aeroplanes, except for the provisions contained in ANO V, No. 24, section 3, respecting the carriage of passengers, section 6, respecting operation in controlled airspace, and section 9, respecting the wearing of helmets;
- the aeroplane may be used for flying training and rental but shall not be used for other commercial purposes;
- (d) the aeroplane shall be operated only by persons holding appropriate licences issued or validated by the Minister of Transport;
- (e) operation in controlled airspace and carriage of passengers other than for flying training is permissible for holders of a Private Pilot Licence-Aeroplane or higher with the appropriate class rating (single engine/land/sea);
- (f) approval from a foreign Aviation Authority is required prior to flight over its territory;
- (g) the aeroplane shall be maintained in accordance with the manufacturer's maintenance program;
- (h) all Mandatory Actions shall be completed in the manner and time frame prescribed by the manufacturer;
- the aeroplane shall not be modified without prior permission from the manufacturer;
- a Statement of Conformity shall be issued in respect of the aeroplane and certified by the manufacturer or an Industry Representative, as applicable; and
- (k) a copy of this Exemption and the Statement of Conformity, shall be carried onboard the aeroplane during flight time.

This Exemption continues in effect until the earliest of:

- (a) the date on which any condition specified in the Exemption is not complied with; or
- (b) the date on which this Exemption is cancelled in writing by the Minister of Transport.

Dated at Ottawa June 10 _____, 1991.

D. Spruston Director General Aviation Regulation

	DEPARTMENT OF TRANSPORT INSPORT CANADA AVIATION GROUP AVIATION REGULATION	MINISTÈRE DES TRANSPORT Groupe aviation transports (La règlementation aérien	AMADA
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PRIVATE/PRIVE	STOL CH 701	(AULA)	7-1780
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REVISED LISTING OF AMATEUR-BUILT AIRCRAFT KITS

July 03, 2002

The following is a revised listing of aircraft kits that have been evaluated and found eligible in meeting the "major portion" requirement of Title 14 of the Code of Federal Regulations (14 CFR part 21), specifically section 21.191 (g). This listing is only representative of those kits where the kit manufacturer or distributor requested an evaluation by the Federal Aviation Administration (FAA) for eligibility and **SHOULD NOT** be construed as meaning the kit(s) are FAA "certified", "certificated", or "approved". There are other aircraft kits that meet the "major portion" requirement of section 21.191 (g), but those manufacturers or distributors have not requested an FAA evaluation. The placement of an aircraft kit on this list is **NOT** a prerequisite for airworthiness certification. The primary purpose of this listing is to assist FAA Inspectors/Designees and other interested individuals by eliminating the duplication of evaluations for "major portion" determination when the aircraft is presented for airworthiness certification as an "Amateur-Built Experimental." Kit manufacturers or distributors whose status is unknown are identified (?) and their address deleted.

Additional information and guidance can be found in Advisory Circular (AC) 20-27E, *Certification and Operation of Amateur-Built Aircraft*, and in AC 20-139, *Commercial Assistance During Construction of Amateur-Built Aircraft*, or by contacting your local FAA Flight Standards District Office (FSDO) or Manufacturing Inspection District Office (MIDO). Those publications and other information pertaining to amateur-built experimental aircraft are available online @ http://av-info.faa.gov/dst/amateur/.

<u>KIT MANUFACTURER</u>	MODEL EVALUATED	DATE FOUND <u>ELIGIBLE</u>
Aircraft Manufacturing & Development Co. (AMD) P.O. Box 639 Eastman, GA 31023	Zodiac CH 640 Kit Assembly Manual - Zodiac CH 640, Rev. 0	05/16/01
Zenair, Ltd. Huronia Airport Midland, Ontario Canada L4K 4K8	CH-200 Parts List dated 10/29/79	10/29/79
	CH-300 Parts List dated 01/01/89	01/13/89
	CH-701 Parts List dated 01/01/89	01/13/89
Zenith Acft. Co. Box 650 Mexico, MO 65265-0650	Zodiac CH 601 Parts dated 01/04/93	05/11/93
	Zodiac CH 601HD Parts list dated 01/04/93	05/11/93

Zodiac CH 601HDS	05/11/93
Parts List dated	
01/04/93	
Zodiac CH 601 XL	03/28/01
Drawings & Manuals	05/20/01
List dated 03/27/01	
STOL CH 701	05/10/93
Parts List dated	
10/09/92	

STOL CH 801 Construction Manuals List dated 03/27/01 03/28/01

END OF LISTING

July 03, 2002

UNITED STATES OF AMERICA DEPARTMENT OF TRANSPORTATION - FEDERAL AVIATION ADMINISTRATION SPECIAL AIRWORTHINESS CERTIFICATE											
Α	PURPOSE OPERATING AMATEUR-BUILT AIRCRAFT										
в	MANU- FACTURER										
с	FLIGHT	FROM NA									
C	FLIGHT	TO NA									
D	N-6128	Q		SERIAL NO. 7 3707							
-	BUILDER D		BBERT	MODEL CH 70							
Ε	DATE OF IS		EXPIRY	ITED							
		LIMITATION	S DATED 3-30-9								
	SIGNATEDE-OF		E- (UHA)	-	DESIGNATION OR OFFICE NO.						
		CE-NEF			DARF501247CE						
Any impl ACC	Any alteration, reproduction or misuse of this certificate may be punishable by a fine not exceeding \$1,000 or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.										
FAX FORM 1130-7 (1012) SEE REVERSE SIDE											
REGISTRATION MOT TRANSFERABLE											
	PARTMENT O		D STATES OF AMERICA	TION ADMINISTRATION	This certificate						
L			OF AIRCRAFT REGIS		must be in the air- cruft when operated.						
NAT	TIONALITY A	ND NI		AIRCRAFT SERIAL NO.							
REG	ISTRATION N	MARKS N 6128	30	73707	3707						
MA	NUFACTURE	R AND MAN	UFACTURER'S DESIG	NATION OF AIRCRAFT	r						
	DUBBERT RO		CH 701	STOL							
L-	CAO Aircrat	ft Address C	ode: 51774653								
					This certificate is issued for registra-						
Ŀ.		T ROGER OCKINGBIRD	00		tion perposes only						
s	1405 M MEXICO	and is what a certif- licate of title.									
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5					at butween private						
Ľ		persons.									
Щ	INDIVIDUAL										
It is certified that the above described aircraft has been extered on the register of the Federal											
International Civil Aviation dated December 7, 1944, and with the Federal Aviation Act											
of 1958, and regulations issued thereunder.											
DA	March 2	Federal Aviation									

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AC Form 8050-3(8/97) Supersedes previous editions

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A	CATEGORY/DESIGNATION Experimental									
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с	FLIGHT TO N/A									
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	DUBBERT R	OGER	NUFACTURER'S D Z00 Code: 53257271	ESIGNATION O LAC CHE01XL	F AIRCRAFT					
- SSUED TO	DUBBE 1405 MEXIC	This certificate is issued for registra- tion purposes only and is not a certifi- icate of sitle. The Federal Avia- tion Administration does not determine rights of ownership as between private								
		persons.								
Aniat	tion Administrat national Civil 4	ion, United States	ed aircraft has been of of America, in accord ember 7, 1944, and w	ance with the Canvo	nation os	U.S. Department of Transportation				
DAT	TE OF ISSUE March	Federal Aviation								

AC Form 8050-318/975 Supersedes previous editions